ALTHEN
SENSORS & CONTROLS

DMU41-01

High Performance MEMS Inertial Measurement Unit

FEATURES

- Tactical Grade 9-DOF MEMS Inertial Measurement Unit
- Silicon Sensing's class leading VSG3QMAX inductive gyro and dual-axis capacitive accelerometer MEMS
- Excellent Bias Instability and Random Walk
  Angular - 0.1°/hr, 0.02°/√hr
  Linear - 15µg, 0.05m/s/√hr
- Non-ITAR
- Compact and lightweight - 50.5 x 50.5 x 51.0 (mm), <178g
- Internal power conditioning to accept 5V to 32V input voltage
- External synchronisation (PPS external trigger)
- RS485 & RS422 interface and sync pulse output
- -40°C to +85°C operating temperature range
- Low power consumption <1.8W at 20°C
- Sealed aluminium housing (IP67)
- Designed to support RTCA/DO-178/DO-254/DO-160G certification
- RoHS compliant
- In-house manufacture from MEMS fabrication to IMU calibration
- Evaluation kit and integration resources available
- First class customer technical support
- User configurable interface

APPLICATIONS

- MEMS alternative to FOG/RLG IMUs
- Guidance, navigation and control (Space, Air, Land, Sea, Subsea)
- Platform/camera/antenna stabilisation
- GPS drop-out aiding
- GNSS (Global Navigation Satellite System)
- Airborne, land-based and hydrographic survey and mapping
- AHRS (Attitude and Heading Reference System)
- INS (Inertial Navigation Systems)
- Small satellite stabilisation and attitude control
- UAVs and ROVs
- Machine control and motion measurements
- Launch vehicle

Version | 12.2025

1 GENERAL DESCRIPTION / 1.1 OVERVIEW

DMU41 is a class leading high performance ninedegree of freedom Inertial Measurement Unit (IMU). It represents the next generation of a family of High Performance IMUs that incorporate an optimised suite of Silicon Sensing's unique resonating ring gyroscopes and capacitive accelerometers, as well as three high performance magnetometers.

DMU41 fuses the outputs from three inductive and three piezoelectric (PZT) resonating ring gyroscopes, to provide three orthogonal measurements of angular rate. The DMU41 also fuses outputs from two independent accelerometers along each axis to provide three orthogonal measurements of linear acceleration.

The unique multi-sensor architecture enables the sensor outputs to be optimally blended to achieve benchmark, all-MEMS inertial performance, providing a realistic alternative to established FOG/RLG based IMUs. DMU41 provides exceptional angle random walk and bias instability coupled with low noise characteristics.

DMU41 has been designed specifically to meet the growing demand for high-end applications requiring a 'tactical grade' IMU without being ITAR controlled. Each DMU41 is individually calibrated in each axis (linearly and angularly) over the full operational temperature range using Silicon Sensing's in-house state-of-the-art test facility.

Silicon Sensing Systems is a market leader in silicon MEMS gyroscopes, accelerometers and inertial measurement systems, specialising in high performance reliability and affordability. Silicon Sensing has a strong heritage in inertial sensing that can be traced back over 100 years. All sensors are based on in-house patented designs which are produced in its own state of the art MEMS foundry. Over 29 million sensors have been delivered to thousands of satisfied customers worldwide, and Silicon Sensing continues to drive performance through technical expertise and continuous innovation.

1.2 INERTIAL SENSORS

The inertial sensing core of DMU41 has two gyros and two accelerometers on each of the three principal axes. Refer to Figure 1.1 DMU41 Functional Block Diagram.

Angular rate on each of the X, Y and Z IMU axes is sensed by a combination of one CRH03 high-precision Inductive-MEMS gyro and one PinPoint® CRM-series high dynamic PZT-MEMS gyro. Six gyros in total.

Linear acceleration on each of the X, Y and Z IMU axes is sensed by three dual-axis, high-integrity Gemini® MEMS accelerometers, effectively providing two accelerometers per axis. Having two independent MEMS accelerometers measuring linear acceleration means that common-mode errors can be largely eliminated resulting in higher performance acceleration measurement.

The IMU software includes 'blending' algorithms to combine the outputs of the multiple sensors per axis in order to achieve higher motion sensing performance and integrity.

The low-bias instability CRH03 gyro has a dynamic measurement range of 200°/s to allow better measurement sensitivity within the normal motion of the host system, yet by utilising the PinPoint® CRM gyro the IMU is capable of operating at up to 490°/s to allow excursions from the normal motion without loss of data.

More information on the sensor operating principles can be found in Section 12.

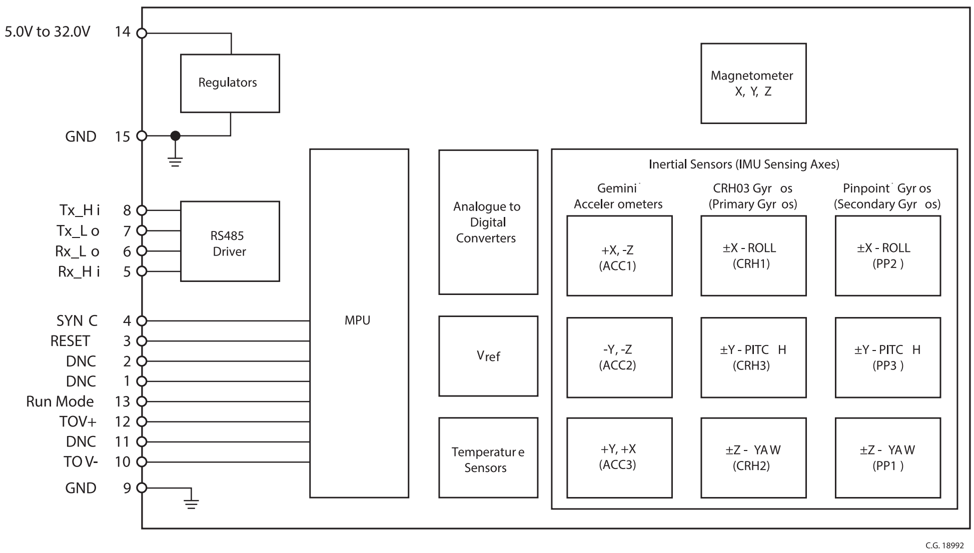
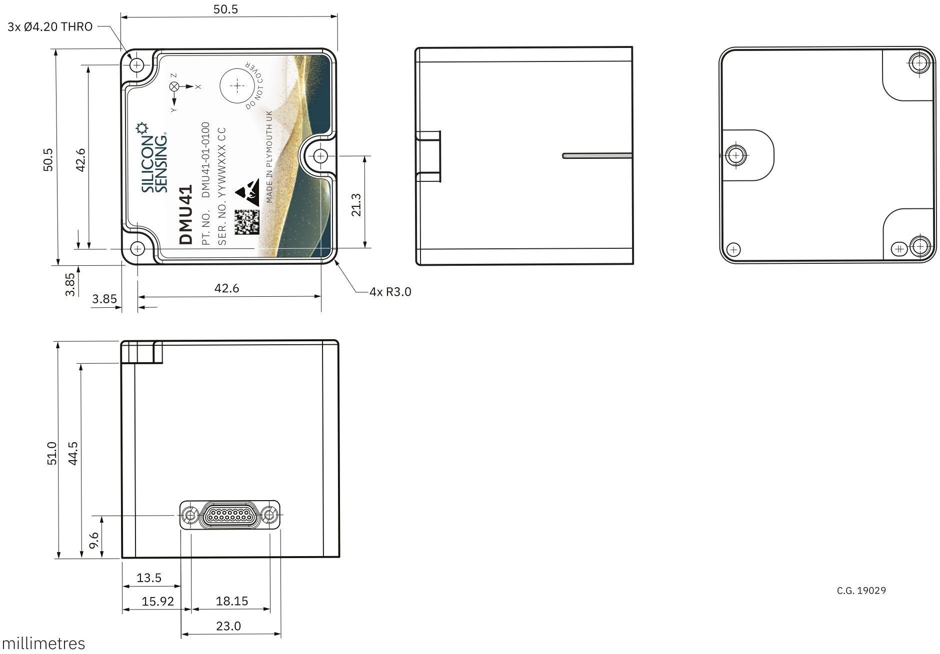


Figure 1.1 DMU41 Functional Block Diagram

OUTLINE CONSTRUCTION (UNIT: mm)

Figure 1.2 DMU41 Unit Overall Dimensions



All dimensions in millimetres
External dimensions are MAX

ORDERING INFORMATION

Item	Description	Overall Dimensions	Part Number
 DMU41-01 IMU	High Performance MEMS Inertial Measurement Unit (USER CONFIGURABLE interface version)	50.5 x 50.5 x 51.0 mm	DMU41-01-0100
 DMU41 Evaluation Kit	Customer Evaluation Kit (EVK) comprising an RS422 to USB Connector, USB Driver and Data Logging Software, Cables and Connectors (DMU41 is NOT included)	Not Applicable	DMU41-00-0500
 DMU41 Mating Connector	Mating connector plug and cable for DMU41	Cable length 450 mm	Glenair MWDM2L-15P-6E5-18 or equivalent

SPECIFICATION

Parameter	Minimum	Typical (1σ) unless stated otherwise	Maximum	Notes
Angular (Roll, Pitch, Yaw)				
Dynamic Range (°/s)	-490	-	+490	Clamped at ±495°/s during over-range
Scale Factor Error (ppm)	-700	±240	+700	Over life
Scale Factor Non-Linearity Error (ppm)	-500	±170	+500	Up to ±200°/s
	-1500	±500	+1500	Between ±200°/s and ±490°/s
Bias (°/hr)	-20	±7	+20	Over operating temperature range. Factory fresh test and during warranty period
Bias Instability (°/h)	-	< 0.1 (Mean)	< 0.2	As measured using the Allan Variance method.
Random Walk ($°/\sqrt{h}$)	-	< 0.02 (Mean)	0.04	
Bias Repeatability (°/h)	-50	±17	+50	Bias Repeatability = $\sqrt{(Bias_{warmup})^2+(Bias_{toto})^2+(Bias_{ageing})^2+(Bias_{temperature})^2}$
Gyro Cross Coupling (%)	-0.4	±0.13	+0.4	Over operating temperature range
Sensor Level Bandwidth (Hz)	-	150	-	-3dB point
IMU Level Bandwidth (Hz)	> 77	-	-	-3dB point
Noise (°/s rms)	-	0.05	0.1	Wideband noise to 100Hz
VRE (°/s/g^2 rms)	-0.006	±0.002	+0.006	4.2g rms stimulus 20Hz to 2,000Hz
g Sensitivity (°/hr/g)	-0.1	±0.033	+0.1	Tested over ±10g

SPECIFICATION CONTINUED

Parameter	Minimum	Typical (1σ) unless stated otherwise	Maximum	Notes
Linear (X, Y, Z)				
Dynamic Range (g)	-10	-	+10	Clamped at ±11.0g during over-range
Scale Factor Error (ppm)	-500	±170	+500	Maximum error at ±1g
Scale Factor Error (ppm) Over Life	-1000	±333	+1000	Maximum error at ±1g. Over life
Scale Factor Non-Linearity Error (ppm)	-5000	±1700	+5000	Maximum error from best straight line calculated at ±1g (over ±10g range)
Bias (mg)	-5.00	±1.70	+5.00	Over operating temperature range. Factory fresh test and during warranty period
Bias Instability (µg)	-	15	30	As measured using the Allan Variance method.
Random Walk (m/s/√h)	-	0.05	0.06	
Bias Repeatability (mg)	-7	±2.3	+7	Bias Repeatability = $\sqrt{(Bias_{warmup})^2+(Bias_{toto})^2+(Bias_{ageing})^2+(Bias_{temperature})^2}$
Acc Cross Coupling (%)	-0.40	±0.13	+0.40	Over operating temperature range
Sensor Level Bandwidth (Hz)	-	250	-	-3dB point
IMU Level Bandwidth (Hz)	>77	-	-	-3dB point
Noise (mg rms)	-	0.9 (Mean)	1.4	Wideband noise to 100Hz
VRE (mg/g² rms)	-0.15	±0.05	+0.15	4.2g rms stimulus 20Hz to 2,000Hz when measured with zero g background acceleration
Temperature Output				
Range (°C)	-45	-	+100	Note that this exceeds operational temperature range
Accuracy (°C)	-	±2	-	Represents the internal DMU41 temperature

ENVIRONMENT, POWER AND PHYSICAL / 4.1 NORMAL OPERATION

Parameter	Minimum	Typical (1σ)	Maximum	Notes
Environment				
RTCA/DO-160G	Tested and in compliance with the environmental requirements of DO-160G			
Operating Temperature Range (°C)	-40	-	+85	Full specification
Storage Temperature Range (°C)	-55	-	+100	-
Operational Shock (g)	-	-	95	6ms, half sinewave. Also shock tested in accordance with DO-160G.
Operational Shock (g) (powered survival)	-	-	1000	1.0ms, half sinewave. (Note 3). Also shock tested in accordance with DO-160G.
Operational Random Vibration (g rms)	-	-	4.2	20Hz to 2kHz
Non-Operational Random Vibration (g rms)	-	-	10.6	20Hz to 2kHz
Humidity (% rh)	-	-	85	Non-condensing
Sealing	IP67	-	-	The DMU41 is sealed and tested to IP67
Environmental Protection				
Audio Frequency Conducted Susceptibility (power inputs)	-	Section 18 Category Z	-	RTCA/DO-160G (Note 1)
Induced Signal Susceptibility	-	Section 19 Category ZWX	-	RTCA/DO-160G
RF Susceptibility (radiated and conducted)	-	Section 20 Category S	-	RTCA/DO-160G
Emission of RF Energy	-	Section 21 Category B	-	RTCA/DO-160G (Note 2)

4.1 NORMAL OPERATION CONTINUED

Parameter	Minimum	Typical (1σ)	Maximum	Notes
Electrical and Interface				
Communication Protocol (standard)	-	RS485, RS422	-	Full duplex communication
Data Rate (Hz)	-	200 (default)	-	Options are 1, 5, 10, 20, 50, 100, 200, 1000, 2000
Baud Rate (bps)	-	460800 (default)	-	Options are 115200, 230400, 460800, 921600
Start-up Time (s) (operational output)	-	< 1.0	1.2	Time to operational output
Start-up Time (s) (full performance)	-	-	20	Time to full performance (mounting dependent)
Quiescent Power (Watts)	-	<1.8W (at 20°C)	-	With 120Ω RS485 termination resistor
Supply Voltage (V)	+5	+12	+32	Unit is calibrated at 12V Note that operation at 5V requires a low impedance supply with short interconnects
Physical				
Size (mm)	-	50.5 x 50.5 x 51.0	-	-
Mass (grams)	-	<178 ± 0.5	-	-

Note 1: DMU41 has been tested in accordance with RTCA/DO-160G section 18 category Z. DMU41 is sensitive to frequencies matching the internal sensor operating frequencies which are 13500Hz to 14500Hz, plus the submultiples of 1/2 and 1/3.

Note 2: DMU41 exceeds D0160G Section 21.4 Category B Conducted RF Emissions limits at 300kHz on the 28V supply line. Additional power line conditioning (EMI filtering) may be required to suppress this depending on host system requirements.

Note 3: This is a survival test. Following exposure to High G shock, linear scale factor performance may degrade by a factor of 3.

Note 4: DMU41 is designed for indoor or outdoor use and to survive short-term immersion in water, up-to the IP67 standard. To maintain integrity around the connector, it is essential that the mating connector is a sealed type, or a suitable sealing compound should be applied around the connector. Product is designed to meet IEC 60664-1 Pollution degree 4.

Note 5: The typical in-rush current for a 5V supply is 1.5A (and <1.8A max), with increasing supply voltage the in-rush current decreases. Therefore, for low supply voltages a supply with a low source impedance is required. Also short cables are recommended. Inclusion of an overcurrent protection device rated at 1A should be considered.

Note 6: Voltages should not be applied to any I/O pin when unit is unpowered.

Note 7: Overranging the DMU41 supply voltage may cause significant damage to the IMU.

Note 8: This product has no user serviceable parts and is not designed to be maintained by anyone other than Silicon Sensing.

4.2 ABSOLUTE MINIMUM/MAXIMUM RATINGS

	Minimum	Maximum
Electrical:		
Vdd	Reverse voltage protected	+32V
ESD protection	-	IEC 61000-4-2 with chassis externally connected to 0V
Life:		
Operational life	5 years	-
MTTF (Ground Mobile) for temperature profile shown	> 50000 hours	Temperature profile: 6% @ -40°C 65% @ 25°C 20% @ 60°C 8% @ 80°C 1% @ 85°C
MTTF (Space Flight @ 30°C)	> 160000 hours	-

Note 1: Improper handling, such as dropping onto hard surfaces, can generate every high shock levels in excess of 10000g. The resultant stresses can cause permanent damage to the sensor.

Note 2: Exposure to the Absolute Maximum Ratings for extended periods may affect performance and reliability.

4.3 COMPLIANCE

- AS9100D
- ISO14001
- ISO17025

RoHS

UK REACH

EMC Performance to RTCA/DO-160-G:

- Section 18 Cat Z
- Section 19 Cat ZW
- Section 20.4 Cat S
- Section 20.5 Cat S
- Section 21.4 Cat B
- Section 21.5 Cat B

5 TYPICAL PERFORMANCE CHARACTERISTICS

This section shows the typical performance of DMU41, operating from a 12V power supply.

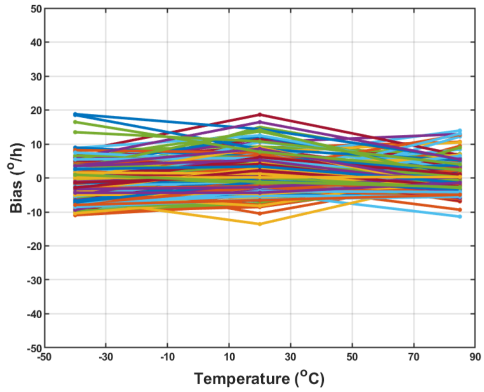


Figure 5.1 Gyro Bias Error (°/h) over Temperature

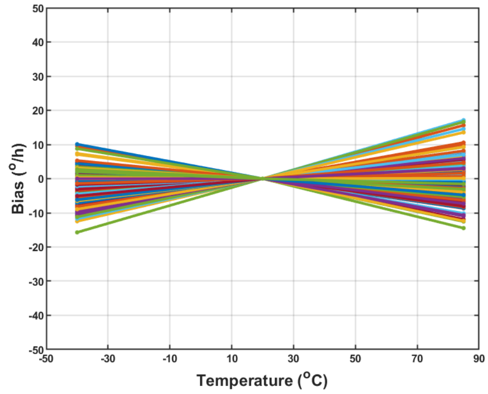


Figure 5.2 Normalised Gyro Bias Error (°/h) over Temperature

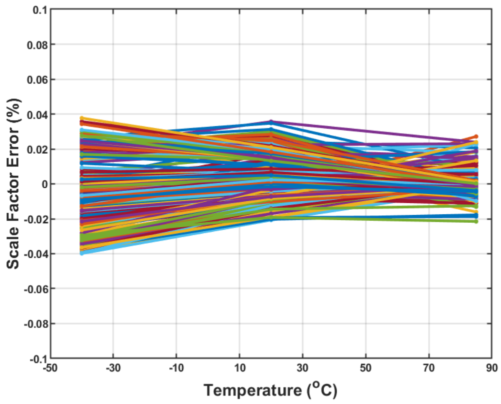


Figure 5.3 Gyro Scale Factor Error over Temperature

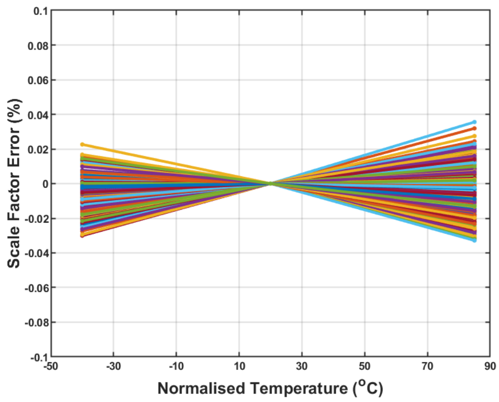


Figure 5.4 Normalised Gyro Scale Factor Error over Temperature

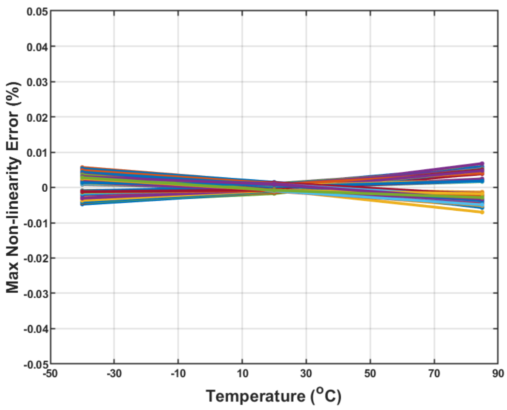


Figure 5.5 Gyro Max Non-Linearity Error (±490°/s range) over Temperature

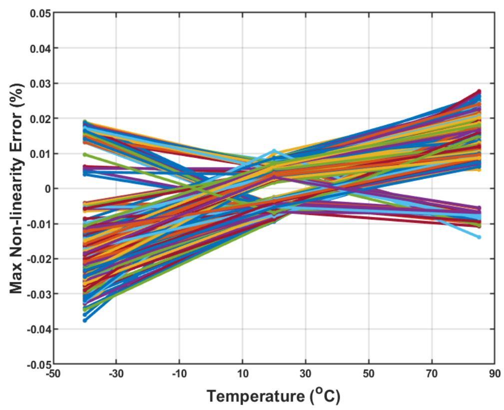


Figure 5.6 Gyro Max Non-Linearity Error (±200°/s range) over Temperature

5 TYPICAL PERFORMANCE CHARACTERISTICS CONTINUED

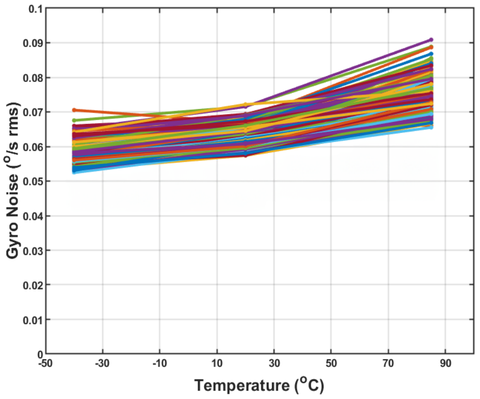


Figure 5.7 Gyro Noise (°/srms) vs Test Chamber Temperature

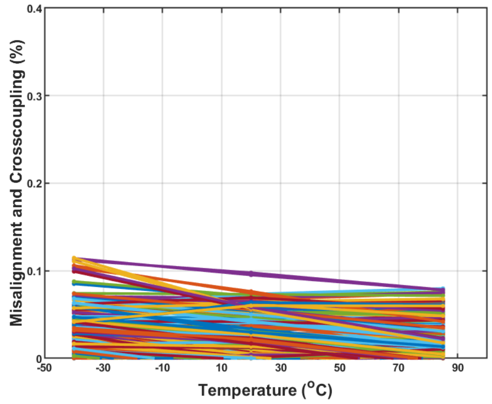


Figure 5.8 Gyro Misalignments and Crosscoupling (±200°/s range) over Chamber Temperature

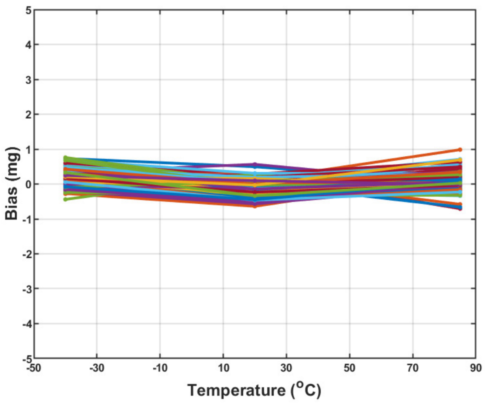


Figure 5.9 Accelerometer Bias Error (mg) over Temperature

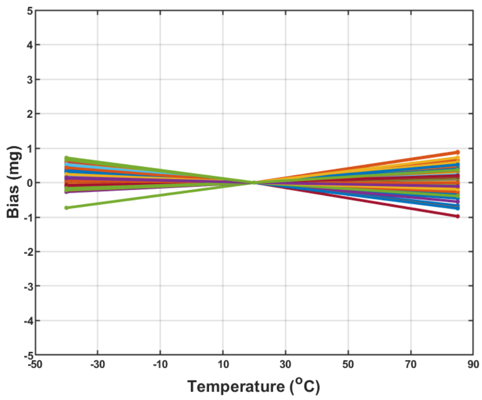


Figure 5.10 Normalised Accelerometer Bias Error (mg) over Temperature

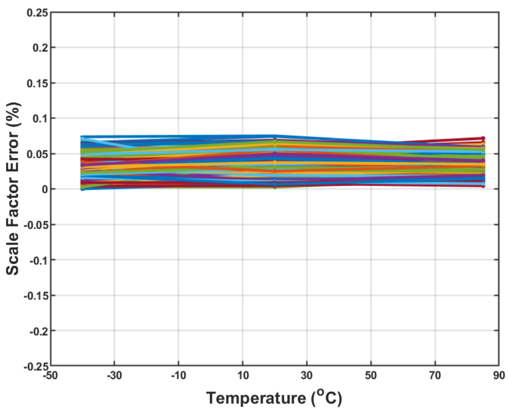


Figure 5.11 Accelerometer Scale Factor Error (±1g range) over Temperature (Plymouth g = 9.81058m/s/s)

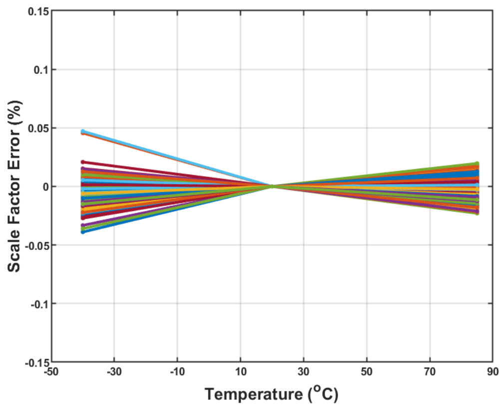


Figure 5.12 Normalised Accelerometer Scale Factor Error (±1g range) over Temperature

5 TYPICAL PERFORMANCE CHARACTERISTICS CONTINUED

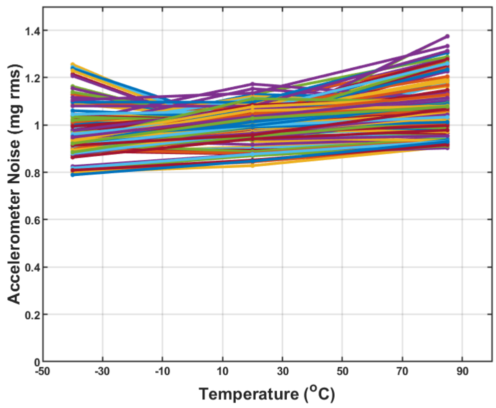


Figure 5.13 Accelerometer Noise vs Test Chamber Temperature

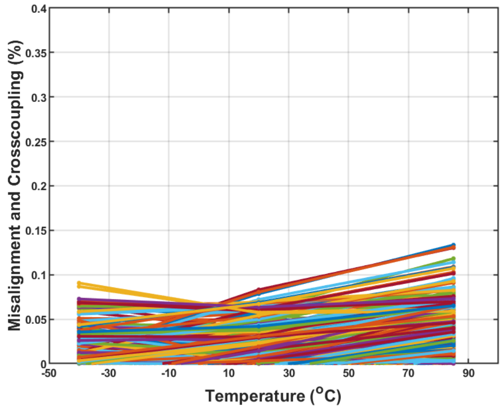


Figure 5.14 Accelerometer Misalignments and Crosscoupling over Temperature

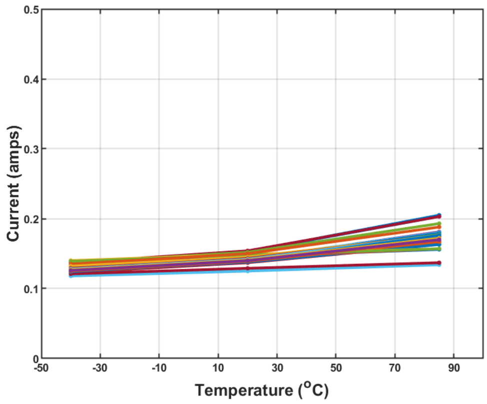


Figure 5.15 Current Consumption vs Chamber Temperature (12V supply)

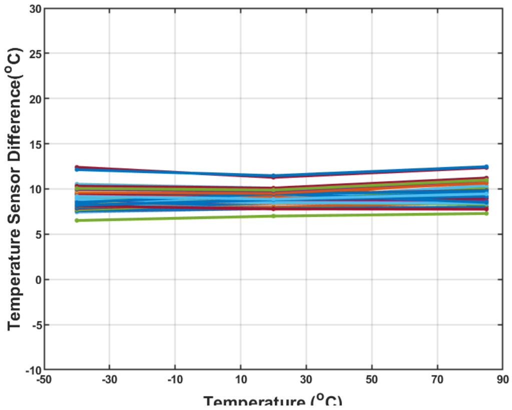


Figure 5.16 DMU41 Temperature Output Difference (°/C) vs Test Temperature (self heating)

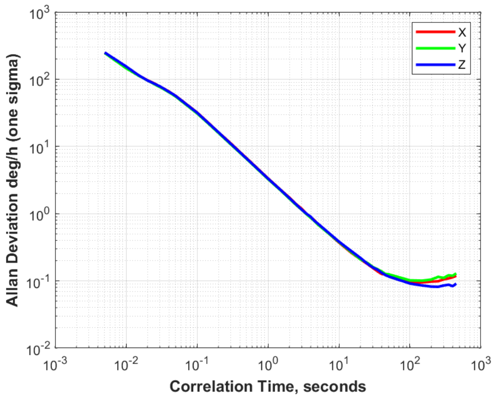


Figure 5.17 Gyro Allan Variance

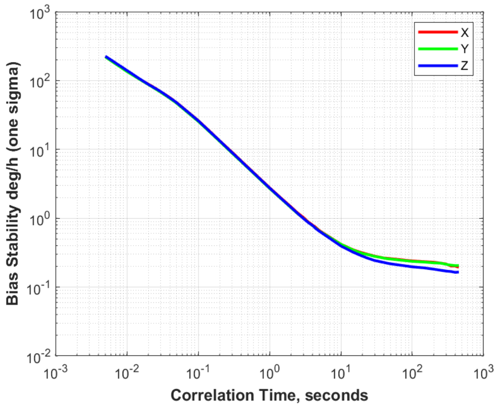


Figure 5.18 Gyro In Run Stability

5 TYPICAL PERFORMANCE CHARACTERISTICS CONTINUED

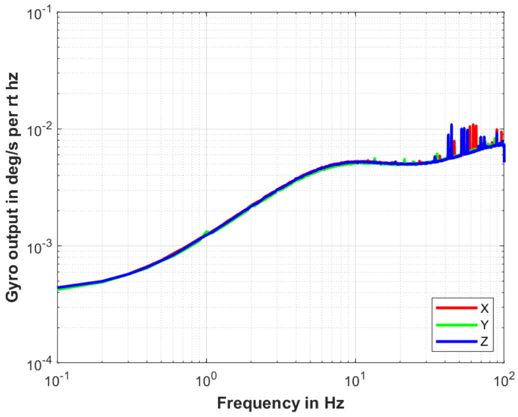


Figure 5.19 Gyro Spectral Data

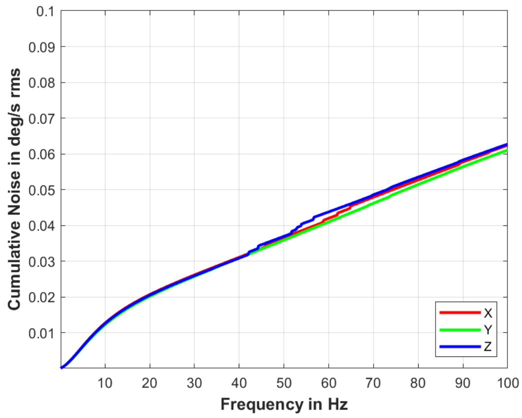


Figure 5.20 Gyro Cumulative Noise

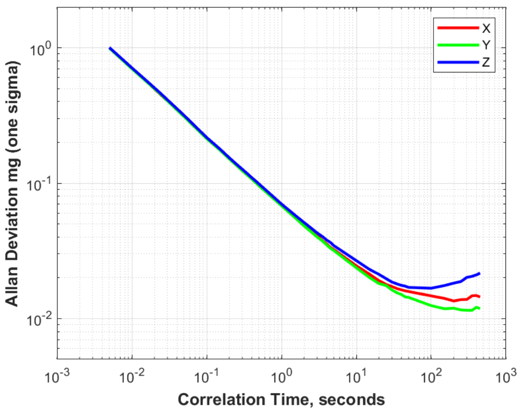


Figure 5.21 Accelerometer Allan Variance

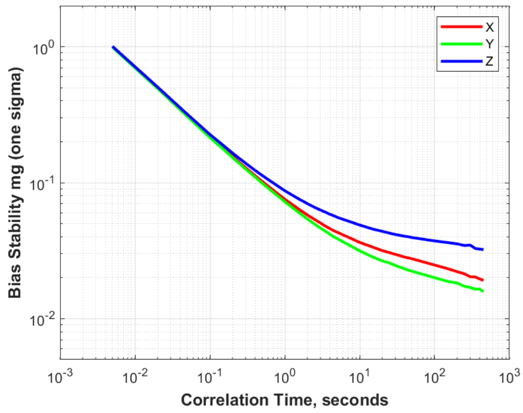


Figure 5.22 Accelerometer In Run Stability

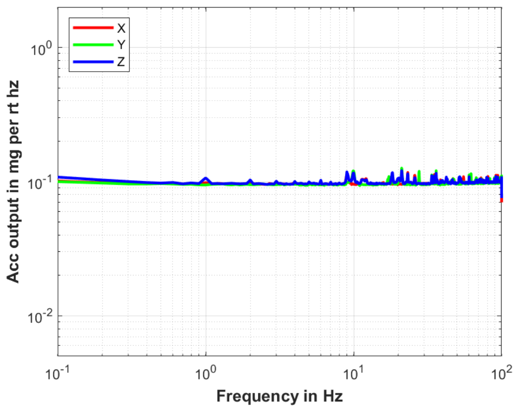


Figure 5.23 Accelerometer Spectral Data

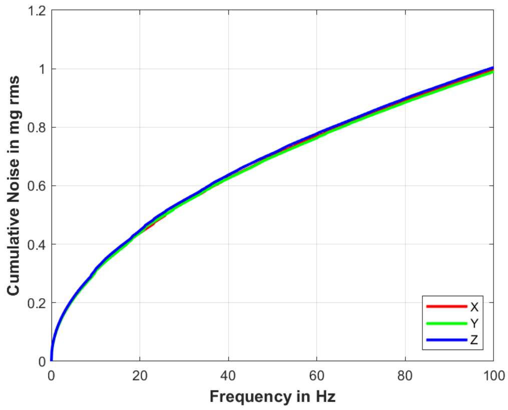


Figure 5.24 Accelerometer Cumulative Noise

6 GLOSSARY OF TERMS

ACC	Accelerometer / Acceleration
AGC	Automatic Gain Control
AHRS	Attitude Heading Reference System
BIT	Built-In-Test
BPS	Bits Per Second
BS	British Standard
CAD	Computer Aided Design
DNC	Do Not Connect
DOF	Degrees of Freedom
DRIE	Deep Reactive Ion Etch
EMC	Electro-Magnetic Compatibility
EMI	Electro-Magnetic Interference
ESD	Electro-Static Damage
EVK	Evaluation Kit
FOG	Fibre Optic Gyro
FP	Fixed Point
GNSS	Global Navigation Satellite System
GPS	Global Positioning System
Hz	Hertz, Cycles Per Second
IMU	Inertial Measurement Unit
INS	Inertial Navigation System
I / O	Input / Output
ITAR	International Traffic in Arms Regulation
MDS	Material Datasheet
MEMS	Micro-Electro Mechanical Systems
MEV	External USB to serial converter
MPU	Microprocessor Unit
MTTF	Mean Time To Failure
NVM	Non-Volatile Memory
PC	Personal Computer
PD	Primary Drive
PLL	Phase Locked Loop
ppm	parts per million
PPS	Pulse Per Second
RAM	Random Access Memory
REACH	Registration, Evaluation, Authorisation and Restriction of Chemicals
RF	Radio Frequency
RLG	Ring Laser Gyro
rms	root mean squared
RoHS	Restriction of Hazardous Substances
RTCA	Radio Tech. Commission for Aeronautics
SD	Secondary Drive
TOV	Time of Validity
UAV	Unmanned Aerial Vehicle
UNC	Unified Coarse
USB	Universal Serial Bus
VCO	Voltage-Controlled Oscillator
VRE	Vibration Rectification Error

7 INTERFACES / 7.1 ELECTRICAL INTERFACE

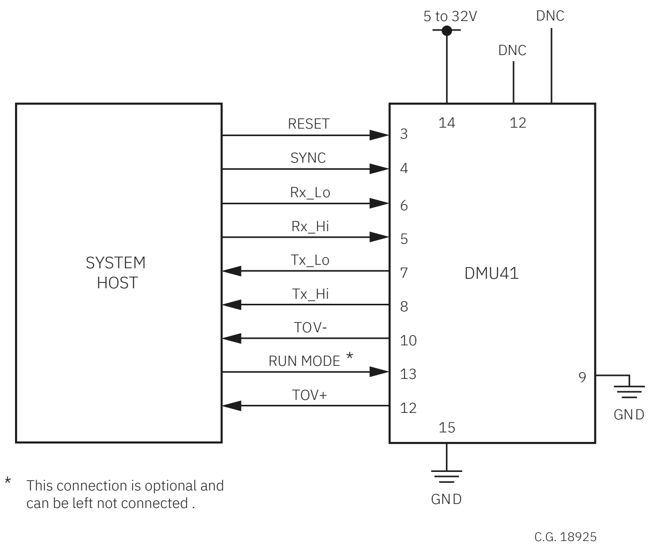


Figure 7.1 Required Connections for RS485 Communications with DMU41

7.2 ELECTRICAL CONNECTOR PINOUT

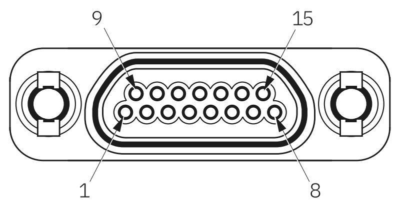


Figure 7.2 DMU41 Socket Connector (Top View)

7.3 CONNECTOR SPECIFICATION

DMU41 uses a 15-way socket connector which is themicro-miniature 'D' type range of connectors, produced by Cinch, Glenair and others.

The DMU41 plug mating connector is a 15 way plug, for example DCCM-15P (DCCM-15P6E518).

Silicon Sensing can supply a mating plug and cable to interface to DMU41 or they are available from electronic component distributors. The part is available from RS, Stock No: 612-6489.

2-56 UNC Female Jackposts are used on the DMU41 connector. 2-56 UNC Jackscrews should be used for connecting to this. A kit is available from RS, Stock Nº: 719-5928.

7.4 PIN INFORMATION

Pin	Label	Signal	In/ Out	Nominal Range	Absolute Max
1	DNC	DNC	DNC	DNC	DNC
2	DNC	DNC	DNC	DNC	DNC
3	RESET	Processor Reset Signal	I	0 to 3.3 V	4 V
4	SYNC	Sync mode input	I	0 to 3.3 V	3.8 V
5	Rx_Hi	+ve Rx RS485 / RS422	I	0 to 3.3 V	3.6 V
6	Rx_Lo	-ve Rx RS485 / RS422	I	0 to 3.3V	3.6 V
7	Tx_Lo	-ve Tx RS485 / RS422	O	0 to 3.3 V	3.4 V
8	TX_Hi	+ve Tx RS485 / RS422	O	0 to 3.3 V	3.4 V
9	GND	Power Return for the DMU41	-	-	-
10	TOV-	Differential Time of Validity	O	0 to 3.3 V	3.4 V
11	Factory Use	Used by SSSL for programming purposes and should not be connected	N/A	0 to 3.3 V	9 V
12	TOV+	Differential Time of Validity	O	0 to 3.3 V	3.4 V
13	Run Mode	-	I	0 to 3.3 V	5 V
14	+Volts	Input voltage to the DMU41 can be between 5V and 32V with respect to GND	-	-	5 V to 32 V
15	GND	Power Return for the DMU41	-	-	-

Table 7.1 Pin Information

7.5 IMU MODE STATES

DMU41 has two IMU mode states: Operational Mode State and Command Mode State. The normal mode state during usage in the host application is the Operational Mode State, and this is the default state during every IMU power up. In the Operational Mode State DMU41 performs and communicates to the host system as per the user configured settings applied while in the Command Mode State.

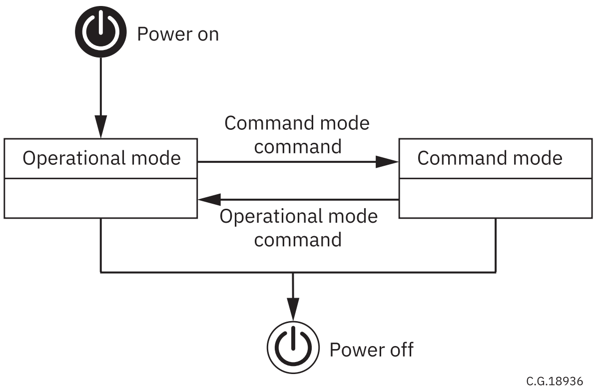


Figure 7.3 DMU41 IMU States

NOTE: As stated in section 4. Voltages should not be applied to any I/O pin when unit is unpowered.

In the Operational Mode State DMU41 performs and communicates to the host system as per the user configured settings applied while in the Command Mode State.

In the Command Mode State the user can configure the operation of DMU41 via a set of ASCII commands, using the utility or by sending strings directly from an external host.

The run mode pin is no longer used for transition into command mode. This pin is a suppress pin only that suppresses serial output in operational mode only.

For full details describing how to set the desired configuration please refer to the Interface Control Document or Evaluation Kit User Manual (DMU41-00-0100-910).

The configurable options are:

1. Set Silicon Sensing Message Endian: Big Endian (default) Little Endian
2. Set Linear Acceleration Output Units (g, mg, or m/s²).
3. Set Angular Rate Output Units (°/s, °/hr, rad/s or rad/hr).
4. Set Silicon Sensing Message Content and Order (see Table 7.2 for more details).
5. Set Output Message Rate (1Hz, 5Hz, 10Hz, 20Hz, 50Hz, 100Hz, 200Hz, 500Hz, 1kHz and 2kHz).
6. Set RS422 / RS485 Bit Rate (115200, 230400, 460800 and 921600bps).
7. Set RS422 / RS485 Stop Bits (1 or 2 bits).
8. Set RS422 / RS485 Parity (odd, even or none).
9. Set Blending Mode Values (Off, On or Overrange)
10. Set Operating Mode (Streaming, TOV, Reset on Trigger, Sample on Trigger (Inertial Only), Sample on trigger (All), Start on Trigger)
11. Set Magnetometer Operating Mode (Off, On)
12. Set Trigger Edge Reference Point (Rising Edge, Falling Edge)
13. Set Time of Validity Assertion Point(Mid-point, Start)
14. Set Blended Angular Rate Bias Values
15. Set Primary Angular Rate Bias Values
16. Set Secondary Angular Rate Bias Values default (0)
17. Set Linear Acceleration Bias Values
18. Set Magnetic Field User Bias Values
19. Set Magnetic Field User Scaling Values
20. Request Silicon Sensing Message Configuration
21. Request Software Part Number
22. Request Unit Serial Number
23. Request System Configuration
24. Request Hardware Part Number
25. Request Unit Variant Number
26. Run Built in tests
27. Request Software Reset

7.6 OPERATIONAL MESSAGE OUTPUT

Messages from DMU41 are built to a standard message structure as summarised in Table 7.2.

Label	Data Type	Data Width (Bytes)	Data Value	Description
Header	Integer	2	0x55AA	Start of message indicator
Count	Integer	2	0 - 65535	Message count (n)
Body	Dependent on message item (0...n)	0 - 400	Dependent on message item (0...n)	Message item data bytes.
Checksum	Integer	2	0 - 65535	Checksum of message

Table 7.2 DMU41 Output Message Structure

7.6.1 MESSAGE BODY

The order and content of the Body portion of the output message is configurable with the data items listed in Table 7.3. The user can configure the DMU41 output message via the ASCII command interface.
All messages are output using big endian format.

Please refer to the DMU41 Evaluation Kit Manual (DMU41-00-0100-910) for further details on configuring the DMU41 output message.

Label	Identifier (Hex)	Data Type	Data Width (Bytes)	Units	Description
Built_In_Test_Start_Up	0x01	Bits	2	Binary	Built in test start up flags.
Built_In_Test_Operational	0x02	Bits	2	Binary	Built in test operational flags.
Error_Indication_01_16_Flags	0x03	Bits	2	Binary	System error Indication bits 1 to 16. These flags indicate an error with the corresponding output message field, a value of 1 in the 5th bit would indicate an error with the 5th message item. The number of bytes within this field support a maximum of 64 data items in the message, which exceeds the number of available message items.
Error_Indication_17_32_Flags	0x04	Bits	2	Binary	System error Indication bits 17 to 32.
Error_Indication_33_48_Flags	0x05	Bits	2	Binary	System error Indication bits 33 to 48.
Error_Indication_49_64_Flags	0x06	Bits	2	Binary	System error Indication bits 48 to 64.
Error_Indication_65_80_Flags	0x07	Bits	2	Binary	System error Indication bits 65 to 80.

Label	Identifier (Hex)	Data Type	Data Width (Bytes)	Units	Description
X_Angular_Rate	0x20	Float	4	°/s, °/hr, rad/hr	X axis blended angular rate.
Y_Angular_Rate	0x21	Float	4	°/s, °/hr, rad/hr	Y axis blended angular rate.
Z_Angular_Rate	0x22	Float	4	°/s, °/hr, rad/hr	Z axis blended angular rate.
X_Primary_Angular_Rate	0x23	Float	4	°/s, °/hr, rad/hr	X axis primary angular rate.
Y_Primary_Angular_Rate	0x24	Float	4	°/s, °/hr, rad/hr	Y axis primary angular rate.
Z_Primary_Angular_Rate	0x25	Float	4	°/s, °/hr, rad/hr	Z axis primary angular rate.
X_Secondary_Angular_Rate	0x26	Float	4	°/s, °/hr, rad/hr	X axis secondary angular rate.
Y_Secondary_Angular_Rate	0x27	Float	4	°/s, °/hr, rad/hr	Y axis secondary angular rate.
Z_Secondary_Angular_Rate	0x28	Float	4	°/s, °/hr, rad/hr	Z axis secondary angular rate.
X_Linear_Acceleration	0x29	Float	4	g, mg or m/s²	X axis linear acceleration.
Y_Linear_Acceleration	0x2A	Float	4	g, mg or m/s²	Y axis linear acceleration.
Z_Linear_Acceleration	0x2B	Float	4	g, mg, m/s²	Z axis linear acceleration.
X1_Linear_Acceleration	0x2C	Float	4	g, mg, m/s²	Raw linear acceleration from an ac celerometer that comprises the first part of the X axis composite linear acceleration.
X2_Linear_Acceleration	0x2D	Float	4	g, mg, m/s²	Raw linear acceleration from an accelerometer that comprises the second part of the X axis composite linear acceleration.
Y1_Linear_Acceleration	0x2E	Float	4	g, mg, m/s²	Raw linear acceleration from an ac celerometer that comprises the first part of the Y axis composite linear acceleration.
Y2_Linear_Acceleration	0x2F	Float	4	g, mg, m/s²	Raw linear acceleration from an accelerometer that comprises the second part of the Y axis composite linear acceleration.
Z1_Linear_Acceleration	0x30	Float	4	g, mg, m/s²	Raw linear acceleration from an ac celerometer that comprises the first part of the Z axis composite linear acceleration.
Z2_Linear_Acceleration	0x31	Float	4	g, mg, m/s²	Raw linear acceleration from an accelerometer that comprises the second part of the Z axis composite linear acceleration.
X_Delta_Theta	0x32	Float	4	°, rad	X axis delta theta using blended angular rate output.
Y_Delta_Theta	0x33	Float	4	°, rad	Y axis delta theta using blended angular rate output.
Z_Delta_Theta	0x34	Float	4	°, rad	Z axis delta theta using blended angular rate output.

Label	Identifier (Hex)	Data Type	Data Width (Bytes)	Units	Description
X_Primary_Delta_Theta	0x35	Float	4	°, rad	X axis delta theta using primary angular rate output.
Y_Primary_Delta_Theta	0x36	Float	4	°, rad	Y axis delta theta using primary angular rate output.
Z_Primary_Delta_Theta	0x37	Float	4	°, rad	Z axis delta theta using primary angular rate output.
X_Secondary_Delta_Theta	0x38	Float	4	°, rad	X axis delta theta using secondary angular rate output.
Y_Secondary_Delta_Theta	0x39	Float	4	°, rad	Y axis delta theta using secondary angular rate output.
Z_Secondary_Delta_Theta	0x3A	Float	4	°, rad	Z axis delta theta using secondary angular rate output.
X_Delta_Velocity	0x3B	Float	4	g/s, mg/s m/s	X axis delta velocity.
Y_Delta_Velocity	0x3C	Float	4	g/s, mg/s m/s	Y axis delta velocity.
Z_Delta_Velocity	0x3D	Float	4	g/s, mg/s m/s	Z axis delta velocity.
X_Quad	0x3E	Float	4	mV	X axis primary gyro quad.
Y_Quad	0x3F	Float	4	mV	Y axis primary gyro quad.
Z_Quad	0x40	Float	4	mV	Z axis primary gyro quad.
X_Primary_Angular_Rate_Temperature	0x44	Float	4	°C	X axis primary angular rate sensor temperature.
Y_Primary_Angular_Rate_Temperature	0x45	Float	4	°C	Y axis primary angular rate sensor temperature.
Z_Primary_Angular_Rate_Temperature	0x46	Float	4	°C	Z axis primary angular rate sensor temperature.
X_Secondary_Angular_Rate_Temperature	0x47	Float	4	°C	X axis secondary angular rate sensor temperature.
Y_Secondary_Angular_Rate_Temperature	0x48	Float	4	°C	Y axis secondary angular rate sensor temperature.
Z_Secondary_Angular_Rate_Temperature	0x49	Float	4	°C	Z axis secondary angular rate sensor temperature.
X_Linear_Acceleration_Temperature	0x4A	Float	4	°C	X axis linear acceleration temperature.
Y_Linear_Acceleration_Temperature	0x4B	Float	4	°C	Y axis linear acceleration temperature.
Z_Linear_Acceleration_Temperature	0x4C	Float	4	°C	Z axis linear acceleration tempera ture.
Housing_Temperature	0x4D	Float	4	°C	Housing temperature.
X_Magnetic_Field	0x4E	Float	4	G	X axis magnetic field.
Y_Magnetic_Field	0x4F	Float	4	G	Y axis magnetic field.
Z_Magnetic_Field	0x50	Float	4	G	Z axis magnetic field.
Magnetic_Field_Temperature	0x52	Float	4	G	Magnetic field sensor temperature.

Table 7.3 DMU41 Configurable Message Items

7.6.2 CHECKSUM

16 bit two's complement of the 16 bit sum of the previous data items. The checksum consists of the running sum of the following 16-bit words:

- Header
- Count
- Body, where all 32 bit values are broken into blocks

of two bytes and added to the running sum Once all message items have been summed the two's complement is taken and the resulting value appended to the message.

7.6.3 SYSTEM START-UP BUILT IN TEST FLAGS

During initialisation a number of parameters can raise a fault condition. Initialisation or start-up faults are persistent and represented by a bit in the system start-up built in test flags message item, these flags are the result of evaluating the initialisation built in tests. The individual bit allocations are shown in Table 7.4 below.

Label	Bit	Description
-	16	Reserved.
-	15	Reserved.
-	14	Reserved.
-	13	Reserved.
-	12	Reserved.
-	11	Reserved.
-	10	Reserved.
-	9	Reserved.
-	8	Reserved.
-	7	Reserved.
Default_Interface	6	Default interface loaded.
Microcontroller_Error	5	Microcontroller error.
Parameter_Range_Error	4	Parameter range error.
System_Device_Error	3	System device error.
Parameter_Checksum_Error	2	Parameter checksum error.
Executable Checksum_Error	1	Software Executable Checksum error.

Table 7.4 DMU41 Start-Up Built in Test Flags

7.6.4 SYSTEM OPERATIONAL BUILT IN TEST FLAGS

During normal IMU operation a number of parameters can raise a fault condition. Operational faults are transient and will only be reported against the output message for which the fault was detected. Each fault is represented by a bit in the system operational built in test flags word, these flags are the result of evaluating the operational built in tests. The individual bit allocations are shown in Table 7.5.

Label	Bit	Description
-	16	Reserved.
-	15	Reserved.
-	14	Reserved
-	13	Reserved.
-	12	Reserved.
-	11	Reserved.
-	10	Reserved.
-	9	Reserved
Magnetic_Field_Plausibility_Error	8	Magnetic field plausibility error.
Temperature_Plausibility_Error	7	Temperature plausibility error.
Angular_Rate_Plausibility_Error	6	Angular rate plausibility error.
Linear_Acceleration_Plausibility_Error	5	Linear Acceleration plausibility error.
Operating_Range_Error	4	Operating range error.
Sensor_Operation_Error	3	Sensor operation error.
Message_Transmission_Timing_Error	2	Message transmission timing error.
Operational_Sequence_Timing_Error	1	Operational sequence timing error.

Table 7.5 DMU41 Operational Built in Test Flags

7.7 SYNCHRONISATION

DMU41 supports four different mechanisms for synchronisation with an external host system. Synchronisation can be either a host enforced process via an input trigger signal or a passive process reporting the point at which the sensors were sampled to the host via an output signal.

7.7.1 TRIGGER SIGNAL

An input trigger signal from the host system is expected to be a square wave pulse. The input trigger supports the generic configuration parameters detailed in Table 7.6.

Parameter	Value	Description
Trigger reference	- Rising edge - Falling edge	The reference point of the trigger pulse used to initiate the different synchronisation process is configurable to either the rising or falling edge of the input signal

Table 7.6 General Trigger Synchronisation Parameters

7.7.2 START ON TRIGGER

In the Start on Trigger mode no output messages are transmitted until the input trigger signal reference edge is detected. Once the input trigger signal is detected messages are streamed at the user configured rate. The characteristics of the synchronisation method are detailed in Figure 7.4 and Figure 7.7.

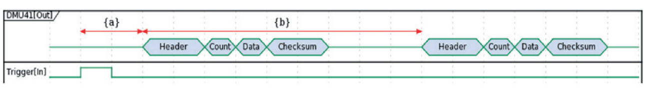


Figure 7.4 Illustration of the Start on Trigger Synchronisation

Parameter	Value	Description
{a} Synchronisation lag time	TBD	The time taken between the trigger signal reference edge and the start of the output message
{b} Output message rate	- 1Hz - 5Hz - 10Hz - 20Hz - 50Hz - 100Hz - 200Hz - 500Hz - 1kHz - 2kHz	The output message rate once the trigger signal has been detected

Table 7.7 Start on Trigger Synchronisation Characteristics

7.7.3 SAMPLE ON TRIGGER (FAST AND SLOW SENSORS)

‚Fast' sensors have a sampling bandwidth of 1kHz (accelerometers and gyroscopes).

'Slow' sensors are those which have a sampling bandwidth of less than 1kHz sampling bandwidth (magnetometers).

In Sample on Trigger mode the sensors are sampled and a message output each time the input trigger signal reference edge is detected.

If the Sample on Trigger mode Fast Sensor Samples is selected then only accelerometers and gyroscopes, are read.

If the Sample on Trigger mode Slow Sensor Samples is selected then accelerometers, gyroscopes and magnetometers are sampled each time a trigger pulse is detected. Due to the lower bandwidth of the sensors being sampled the input trigger rate is limited in this mode. The characteristics of the synchronisation method are detailed in Figure 7.5 & Table 7.8.

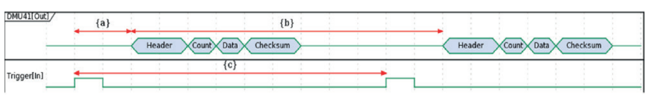


Figure 7.5 Illustration of the Sample on Trigger Synchronisation

Version | 12.2025

Parameter	Value	Description
{a} Synchronisation lag time	- Fast sensor signals TBD - Slow sensor signals TBD	The time between the input trigger signal reference edge and the start of the output message.
{b} Output message rate	Dependent on input trigger period	The output message rate once the trigger signal has been detected
{c} Trigger period	- Periods up to 1kHz for Fast Sensor sampling - Periods up to 10Hz for Slow Sensor sampling	The period between trigger pulses

Table 7.8 Sample on Trigger Synchronisation Characteristics

Parameter	Value	Description
{a} Output message rate	- 1Hz - 5Hz - 10Hz - 20Hz - 50Hz - 100Hz - 200Hz - 500Hz - 1kHz - 2kHz	The output message rate
{b} Synchronisation lag time	TBD	The time between the input trigger signal reference edge and the start of the output message

Table 7.9 Reset on Trigger Synchronisation Characteristics

7.7.4 RESET ON TRIGGER

In the Reset on Trigger mode the internal sensor sampling regime is reset each time an input signal is detected. Once the internal sensor sampling regime is reset, output messages are streamed at the user configured rate. The characteristics of the synchronisation method are detailed in Figure 7.6 and Table 7.9.

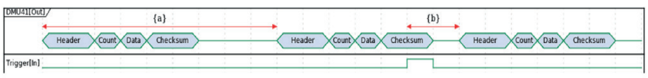


Figure 7.6 Illustration of the Reset on Trigger Synchronisation

7.7.5 TIME OF VALIDITY

In the Time of Validity mode a differential output pulse is output when the sensor sampling regime is initiated. The characteristics of the synchronisation method are detailed in Figure 7.7 and Table 7.10.

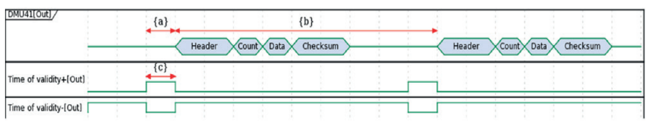


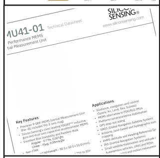
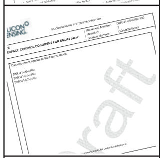
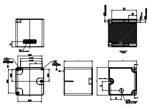
Figure 7.7 Illustration of the Time of Validity Synchronisation

Version | 12.2025

Parameter	Value	Description
{a} Synchronisation lag time	TBD	The time taken from the leading edge of the time of validity output signal and the start of the output message
{b} Output message rate	- 1Hz - 5Hz - 10Hz - 20Hz - 50Hz - 100Hz - 200Hz - 500Hz - 1kHz - 2kHz	The output message rate
{c} Time of validity pulse width	TBD	The width of the output time of validity pulse

Table 7.10 Time of Validity Synchronisation Characteristics

8 DESIGN TOOLS AND RESOURCES AVAILABLE

Item	Description of Resource	Part Number	Order/Download
	**DMU41 Brochure:** A one page sales brochure describing the key features of the DMU41 Inertial Measurement Unit.	DMU41-00-0100-131	Download (https://siliconsensing.com/)
	**DMU41 Datasheet:** Full technical information on all DMU41 Dynamic Measurement Unit part number options. Specification and other essential information for assembling and interfacing to DMU41 Inertial Measurement Unit.	DMU41-01-0100-132	Download (https://siliconsensing.com/)
	**DMU41 ICD:** The interface control document.	DMU41-00-0100-130	Download (https://siliconsensing.com/)
	**Solid Model CAD files for DMU41 Inertial Measurement Unit:** Available in .STP file format.	DMU41-00-0100-403_ Iss_1.stp	Download (https://siliconsensing.com/)
	**DMU41 Installation Drawing:** Drawing containing host interface geometry.	DMU41-00-0100-403	Download (https://siliconsensing.com/)
	**Customer Evaluation Kit (EVK)** comprising an RS422 to USB Connector, USB Driver and Data Logging Software, Cables and Connectors (DMU41 is NOT included)	Not Applicable	DMU41-00-0500
	**Mating connector plug and cable for DMU41**	Cable length 450 mm	Glenair MWDM2L-15P-6E5-18 or equivalent
	**RoHS compliance statement for DMU41:** DMU41 is fully compliant with RoHS. For details of the materials used in the manufacture please refer to the MDS Report.	-	-

8.1 DMU41 EVALUATION KIT (EVK)

The DMU41 Evaluation Kit enables the output data from the DMU41 to be viewed and logged for testing and evaluation purposes.

Figure 8.1 DMU41 Evaluation Kit

8.1.1 DMU41 EVALUATION KIT CONTENTS

The DMU41 Evaluation Kit (part number DMU41-00-0500) contains the following:

- MEV RS485i to USB converter.
- Mating plug and cable.
- User manual (downloadable from http://www.siliconsensing.com).

Note: DMU41 is NOT included in the EVK, it must be purchased separately.

8.1.2 SYSTEM REQUIREMENTS

The DMU41 Evaluation Kit requires a PC with a USB port. The requirements for the PC are as follows:

- Microsoft® Vista®, Windows 7, Windows 8, Windows 10 or Windows 11, Operating Systems. The software has not been tested on any other Operating System and therefore correct functionality cannot be guaranteed.
- Minimum of 500Mb of RAM.
- 500Mb of free hard drive space plus space for logged data (typical data rate ≈ 50kbit/s).
- High power or self-powered USB 2.0 Port.

9 PART MARKINGS

DMU41 is supplied with an adhesive label attached. The label displays readable DMU41 part and serial numbers.

The serial number is a numeric code, YYWWXXXX CC where:

YY = Manufacturing year number
WW = Manufacturing week number
XXXX = Part ID number
CC = Revision

A 4x4 data matrix barcode containing the part identification number is also displayed on the label.

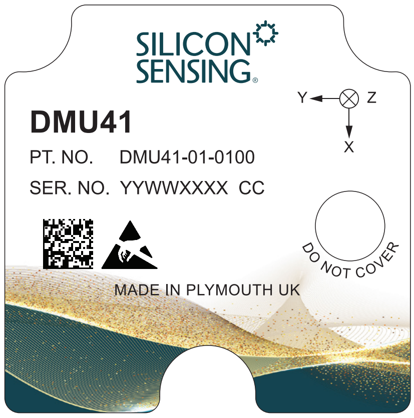


Figure 9.1 DMU41 Label

10 INSTALLATION DETAILS

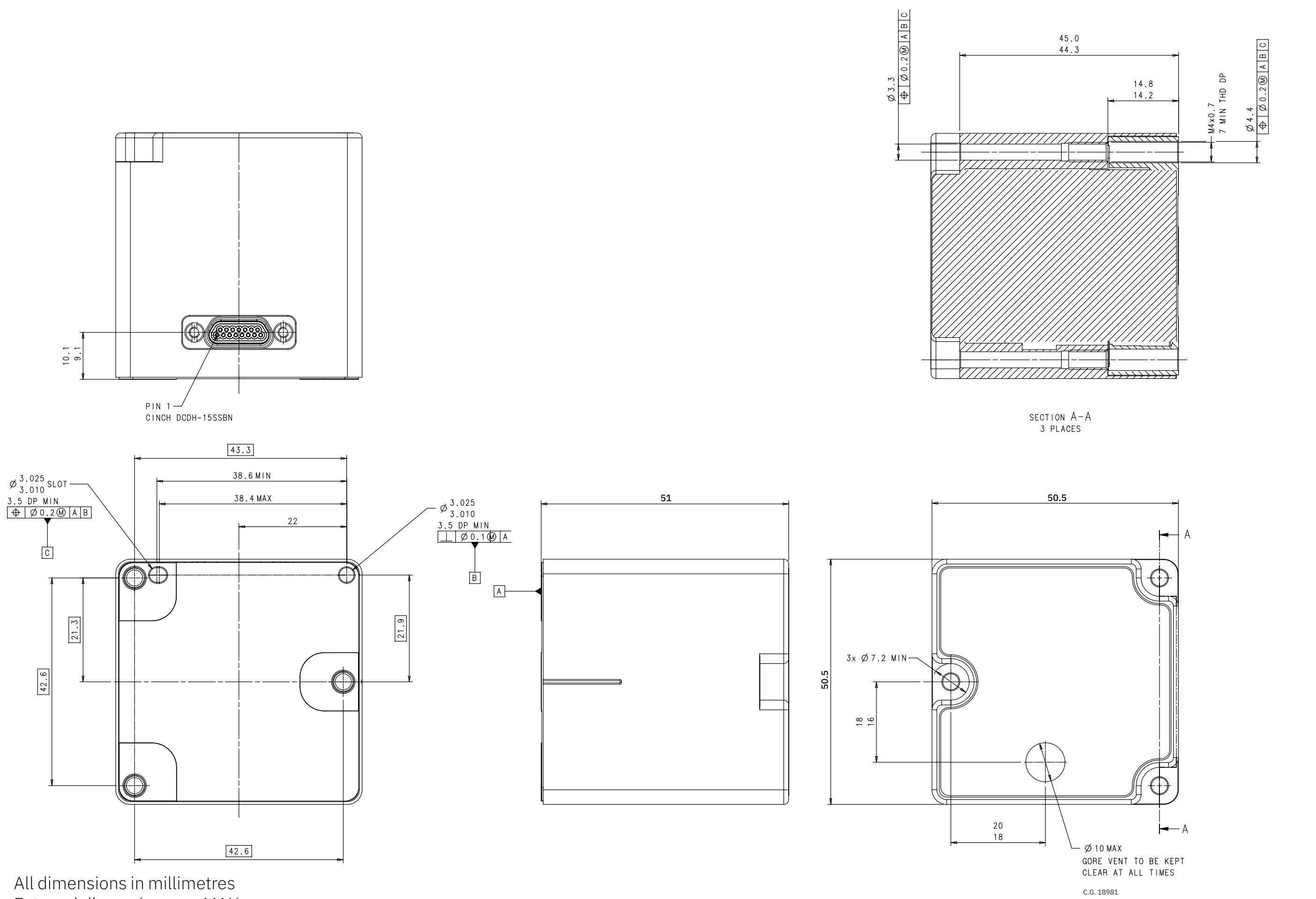
Figure 10.1 shows the installation drawing for the DMU41. DMU41 is designed to accommodate 3-point mounting from either the top (standard) or bottom (bulkhead) of the unit. Standard M3 screws should be used when standard mounting is used, with a minimum length of 50mm. Bulkhead mounting uses M4 screws, where the thread engagement is located 14.4mm down the length of the hole. There is a thread depth of 8.0mm available for thread engagement.

The use of dowel holes is critical to achieve alignment and repeatability. Alignment of the part should be with respect to these, using dowel pins. The dowel holes are designed to be used with two Ø3mm (in accordance with BS EN ISO 8734 or BS EN ISO 2338) dowel pins provided by the host.

The DMU41 mounting screw torque settings will be dependent on the host application; it will for example vary depending on the specification of the screw, the material of the host structure and whether a locking compound is used. The suggested torque setting for securing a DMU41 to an aluminium host structure using steel screws and a thread locking compound is 0.2Nm. This information is provided for guidance purposes only. The actual torque settings are the responsibility of the host system designer. The DMU41 should not be disassembled. This could compromise the calibration and will invalidate the warranty.

Note: The DMU41 housing incorporates a vent to enable a planned future addition of a barometric pressure sensor. This vent is located on the top face of the housing, in the lower right corner of the IMU label. This area is marked 'PLEASE DO NOT COVER'. The vent has a Gore-Tex® seal installed.

To maintain enclosure integrity and IP67 rating, the Gore-Tex® seal must not be damaged, altered or covered in any way.



All dimensions in millimetres
External dimensions are MAX

Figure 10.1 DMU41 Installation Drawing

11 AXIS DEFINITIONS AND SENSING POINTS

The DMU41 uses 6 gyroscopes and 6 accelerometers in a paired configuration to optimise performance for each axis. Figure 11.1 shows the axis definitions for the DMU41.

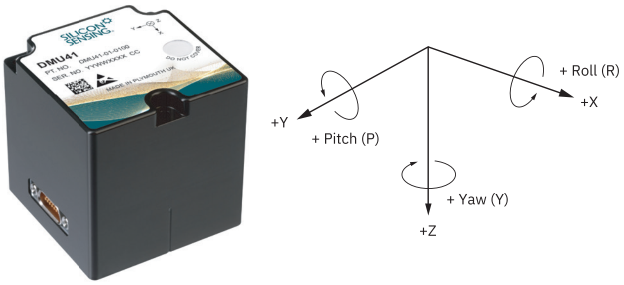


Figure 11.1 Axis Definitions

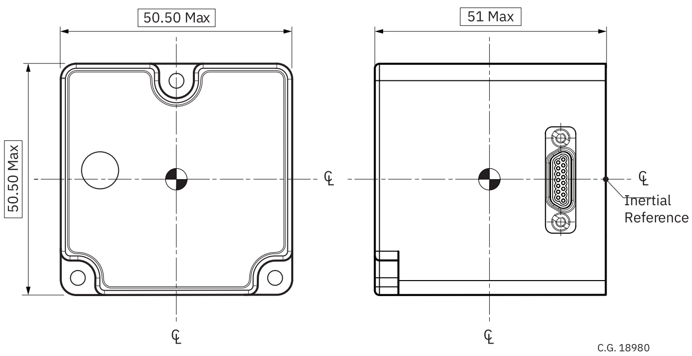


Figure 11.2 Position of the Inertial Reference Point

There are two accelerometers per axis within the DMU41. The sensors used in each axis are orientated to reduce common mode errors and improve noise.
Table 11.1 shows the effective mid-point position of the pairs of accelerometers used for each axis.
Size effect compensation is not carried out within the DMU41 and these values will enable the user to provide external size effect compensation should this become necessary within the application.

Accelerometer	Sensing Element Position (Relative to Inertial Reference Point), mm		
	X	Y	Z
X Accelerometer	+10.35	+0.6	+10.25
Y Accelerometer	+4.85	-2.80	+26.45
Z Accelerometer	-1.05	-9.2	+23.6

Table 11.1 Accelerometer Sensing Positions

12 DMU41 CONSTRUCTION AND THEORY OF OPERATION

12.1 IMU CONSTRUCTION

DMU41 is an aluminium alloy assembly comprising base, housing, sensor block, sensor assemblies and IMU electronics.

The base and housing are sealed using a self-forming gasket and secured by three machine screws to provide a waterproof enclosure. A micro-miniature 'D' type socket connector located on the top face of the housing provides the electrical interface to the host system. The top face of the housing displays the DMU41 part marking information.

DMU41 is aligned to the host system using two Ø3mm dowels in the host platform which locate with matching dowel holes in the bottom face of the base. The standard mounting for DMU41 is secured using M3 machine screws, for alternative bulkhead mounting M4 machine screws are used (see Section 10 - Installation Details).

A precision machined aluminium 3-Axis Sensor Block, secured to the DMU41 Base by machine screws provides accurate alignment and support for the DMU41 MEMS inertial sensor assemblies and IMU electronics. Internally generated heat from the sensor assemblies and IMU electronics is absorbed into the sensor block and surrounding housing and conducted to the host via the base and to the ambient atmosphere via the housing.

12.2 SENSOR CONSTRUCTION AND THEORY OF OPERATION

Silicon MEMS Inductive Ring Gyroscope

The silicon MEMS ring is 6mm diameter by 100µm thick, fabricated by Silicon Sensing Systems using a DRIE (Deep Reactive Ion Etch) bulk silicon process. The ring is supported in free-space by eight pairs of symmetrical 'dog-leg' shaped legs which support the ring from the supporting structure on the outside of the ring.to the ambient atmosphere via the housing.

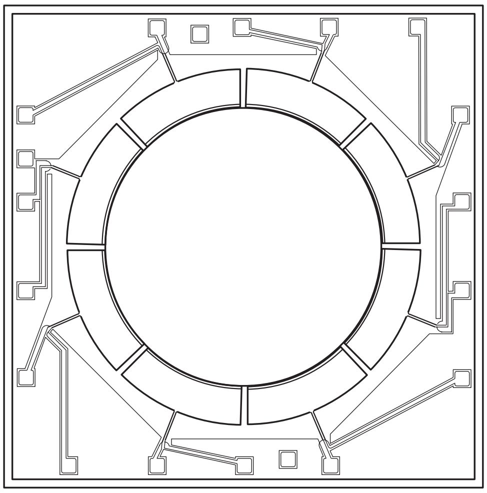


Figure 12.1 Silicon MEMS Ring

The bulk silicon etch process and unique patented ring design enable close tolerance geometrical properties for precise balance and thermal stability and, unlike other MEMS gyros, there are no small gaps to create problems of interference and stiction. These features contribute significantly to DMU41's bias and scale factor stability over temperature, and vibration immunity. Another advantage of the design is its inherent immunity to acceleration induced rate error or 'g-sensitivity'.

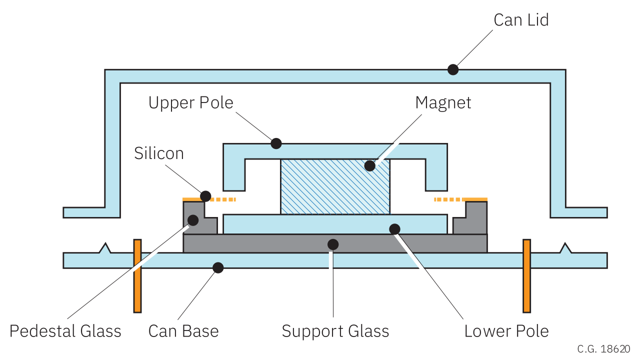


Figure 12.2 MEMS VSG3Q ^{MAX} Sensor

The ring is essentially divided into 8 sections with two conductive tracks in each section. These tracks enter and exit the ring on the supporting legs. The silicon ring is bonded to a glass pedestal which in turn is bonded to a glass support base. A magnet, with upper and lower poles, is used to create a strong and uniform magnetic field across the silicon ring. The complete assembly is mounted within a hermetic can.

The tracks along the top of the ring form two pairs of drive tracks and two pairs of pick-off tracks. Each section has two loops to improve drive and pick-off quality.

One pair of diametrically opposed tracking sections, known as the Primary Drive PD section, is used to excite the cos2q mode of vibration on the ring. This is achieved by passing current through the tracking and, because the tracks are within a magnetic field, causes motion on the ring. Another pair of diametrically opposed tracking sections, known as the Primary Pick-off PP section and are used to measure the amplitude and phase of the vibration on the ring. The Primary Pick-off sections are in the segments 90° to those of the Primary Drive sections. The drive amplitude and frequency is controlled by a precision closed loop electronic architecture with the frequency controlled by a Phase Locked Loop (PLL), operating with a Voltage Controlled Oscillator (VCO), and amplitude controlled with an Automatic Gain Control (AGC) system. The primary loop therefore establishes the vibration on the ring and the closed loop electronics is used to track frequency changes and maintain the optimal amplitude of vibration over temperature and life. The loop is designed to operate at about 14kHz.

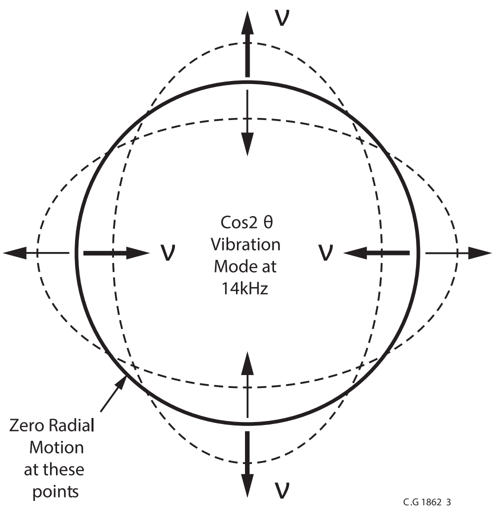


Figure 12.3 Primary Vibration Mode

Having established the cos2q mode of vibration on the ring, the ring becomes a Coriolis Vibrating Structure Gyroscope. When the gyroscope is rotated about its sense axis the Coriolis force acts tangentially on the ring, causing motions at 45° displaced from the primary mode of vibration. The amount of motion at this point is directly proportional to the rate of turn applied to the gyroscope. One pair of diametrically opposed tracking sections, known as the Secondary Pick-off SP section, is used to sense the level of this vibration. This is used in a secondary rate-nulling loop to apply a signal to another pair of secondary sections, known as the Secondary Drive SD. The current applied to the Secondary Drive to null the secondary mode of vibration is a very accurate measure of the applied angular rate. All of these signals occur at the resonant frequency of the ring. The Secondary Drive signal is demodulated to baseband to give a voltage output directly proportional to the applied rate in free space.

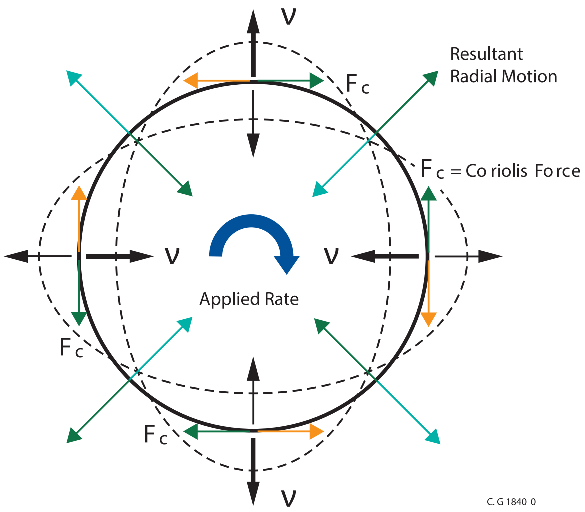


Figure 12.4 Secondary Vibration Mode

The closed loop architecture of both the primary and secondary loops results in excellent bias, scale factor and non-linearity control over a wide range of operating environments and life. The dual loop design, introduced into this new Sensor Head design, coupled with improved geometric symmetry, results in excellent performance over temperature and life. The discrete electronics employed in DMU41 ensures that performance is not compromised.

SILICON MEMS CAPACITIVE ACCELEROMETER

The accelerometer contains a seismic ‘proof mass’ with multiple fingers suspended via a ‘spring’ from a fixed supporting structure. The supporting structure is anodically bonded to the top and bottom glass substrates thereby fixing it to the sensor package base.

When the accelerometer is subjected to a linear acceleration along its sensitive axis, the proof mass tends to resist motion due to its own inertia, therefore the mass and its fingers become displaced with respect to the interdigitated fixed electrode fingers (which are also fixed to glass substrates). Air between the fingers provides a damping effect. This displacement induces a differential capacitance between the moving and fixed silicon fingers which is proportional to the applied acceleration.

Capacitor plate groups are electrically connected in pairs at the top and bottom of the proof mass. In-phase and out of phase waveforms are applied by the ASIC separately to the ‘left’ and ‘right’ finger groups. The demodulated waveforms provide a signal output proportional to linear acceleration.

Figures 12.5(a) and 12.5(b) provide schematics of the accelerometer structure and control loop respectively.

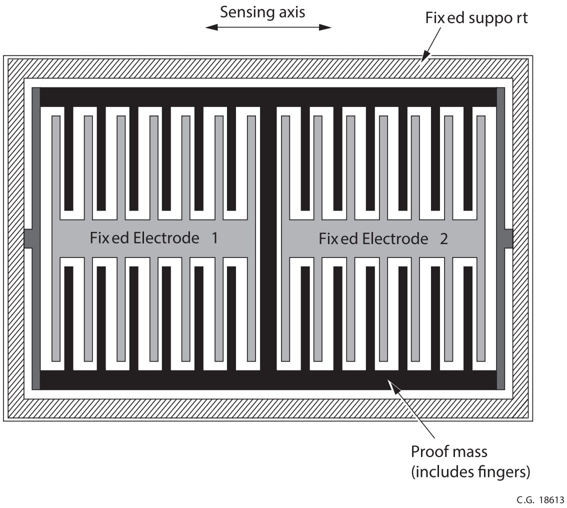


Figure 12.5(a) Schematic of Accelerometer Structure

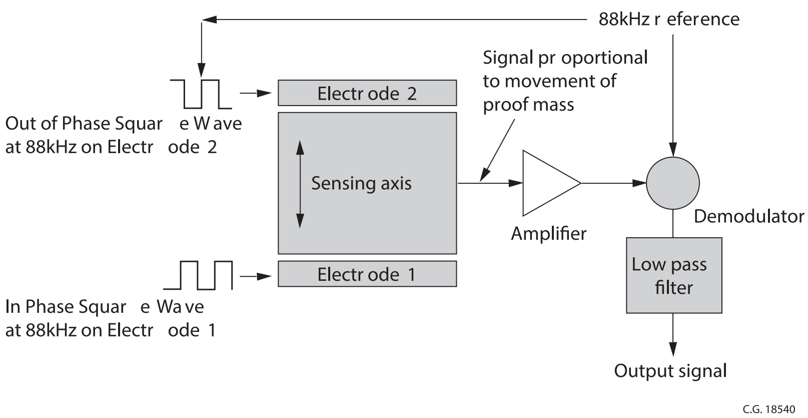


Figure 12.5(b) Schematic of Accelerometer **Control Loop**